Lecture 7 Heat Transfer & Thermal Insulation

J. G. Weisend II

- Introduce conduction, convection & radiation heat transfer as they apply to cryogenics
- § Describe design techniques to reduce heat transfer into cryogenic devices
- § Allow the estimating and scaling of heat leaks into cryogenic devices
- Warning! Not a full description of heat transfer
	- Many topics (boiling, detailed convection calculations, complicated geometries in radiation heat transfer etc) won't be covered
	- Should, however, be a good example of how heat transfer theory can be applied to practical problems.

Three Ways to Transfer Heat

- Conduction
	- Heat transfer through solid material
- Convection
	- Heat transfer via a moving fluid
		- » Natural or free convection motion caused by gravity (i.e. density changes)
		- » Forced motion caused by external force such as a pump
- Radiation
	- Heat transferred by electromagnetic radiation/photons
- § There is no such thing as a perfect insulator though we can design systems with very small heat leaks
- All matter above 0 K radiate heat
	- Remember we can't get to 0 K $3rd$ Law of Thermodynamics though we can get vanishingly close
- Heat flows from high temperature to low
	- Heat leaks in, cold doesn't leak out

$$
Q = -K(T)A(x)\frac{\partial T}{\partial x}
$$

If we assume constant cross section we get:

$$
Q = -A/L \int_{T_C}^{T_H} K(T) dT
$$

- Reduce conduction heat leak by:
	- Low conductivity material: make K(T) small
	- Reduce cross sectional area: make A small
	- Increase length: make L large
	- For a given T_c make T_H smaller: i.e. use intermediate temperature heat intercepts
		- » You still have heat leak from 300 K to this intermediate temperature but remember Carnot, It's more thermodynamically efficient to remove heat at higher temperatures

Design Example ILC Cryomodule Support Post

- § Total Heat Leak (conduction & radiation)
	- \cdot 70 K 10.5 W
	- $-5K 0.9W$

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- 2 K 0.03 W
- Can support up to 50 kN

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Conduction Heat Transfer

• Conduction heat leaks may be estimated by the use of Thermal Conductivity Integrals (Lecture 4)

$$
Q = -G(\theta_1 - \theta_2)
$$

■ Fundamental Equation: Newton's law of cooling

 $Q = hA(T_{surface} - T_{fluid})$

- where h is the heat transfer coefficient and is a function of Re, Pr, geometry etc depending on the situation
- In cryogenics we eliminate convection heat leak in cryogenic systems by "simply" eliminating the fluid – vacuum insulation
- Using vacuum insulation to create vessels capable of storing cryogenic liquids was first done by James Dewar – who liquefied hydrogen
	- Such vessels are frequently called dewars though not always, more later
	- Thermos bottles are a simple example of this approach

Design Example Vacuum Insulated Test Cryostat

- § Contains 3 Vacuum Spaces
	- 1 between 300 K wall and $LN₂$ bath
	- \cdot 1 between LN₂ bath and LHe bath
	- 1 between LHe bath and experiment

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 \Rightarrow

Vacuum Insulation

Example How much vacuum is enough?

- This of course depends on the heat leak requirements but generally we want to be below 10⁻⁵ torr If we maintain this level or better we can generally neglect the convection heat leak for most applications.
	- » Cryogenic Engineering, Flynn (1997) has a good discussion of calculating heat leak due to residual gas pressure

§ Cryopumping

- At cryogenic temperatures almost all common gases condense and freeze onto the cold surface. Typically, we'll see that once surface are cooled to \sim 77 K the isolation vacuum will drop to the 10^{-8} torr or better range if the system is leak tight and doesn't have significant outgassing
- But don't just start cooling with everything at room pressure » Heat leak will likely be too high
	- » Safety hazards due to enrichment of LOX on cold surfaces
	- » Large amounts of condensed gases in vacuum space can lead to other problems including rapid pressure rise upon warming and possible solid conduction
	- \ast Best practice is to be at least 10⁻³ torr before cooling, lower pressures are better but there may be operational tradeoffs

Outgassing and Getters

- All material outgas into a vacuum. This can raise the pressure in a sealed vacuum space
- Reduce outgassing by:
	- Minimize amount of polymers, wire insulation, FRP etc difficult
	- Keep vacuum surfaces as clean as possible. Remove any oil or cutting fluid, wear gloves etc.
- § Getters: materials inserted into vacuum spaces to remove residual gas at low pressures
- In cryogenic systems, getters may be useful in removing residual gas and passively managing small leaks

Outgassing and Getters

■ 3 types of getters

- Adsorbers –gas bonds to surface
	- » Activated charcoal, silica gel
	- » Effectiveness increases with decreasing temperature good for cryogenic systems
- Chemical getters chemical reaction between material and gas » Ba & other Alkali metals – not very common in cryogenics
- Solution or absorber getters gas is absorbed in interstitial space of metals » Ti, Zr, Th works well with H_2 , O₂ and N₂
	- » Much better at room temperature
	- » Occasional use in room temperature applications in cryogenic systems

- § It turns out that one of the most common and effective materials used for getters of low pressure He gas is activated charcoal made from coconut husks.
- There is a significant amount of this material in the LHC magnet cryostats

The Professor says: "Lets look for the Higgs!"

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- Not all cryogenic systems use vacuum insulation
- § This is particularly true of storage vessels for fluids other than helium
- Reasons for using alternatives to vacuum insulation
	- Cost
	- Weight Space shuttle main tank
	- Required hold time related to size
	- Complex vessel shapes
- § Some solutions
	- Expanded closed or open cell foams
	- Rock wool, fiberglass or other porous material
- § These all require vapor barriers to prevent air from being pulled into the insulation and condensed (can cause both a safety hazard via $O₂$ enrichment & reduce effectiveness)

Design Example: Complex Foam Insulation System: LH₂ Tank for 2nd Stage Saturn V

- Allows helium purging of the insulation
- Weight \sim 4.15 kg/m²
- § Performance: measured effective thermal conductivity (0.86 1.1 mW/ cm K) at T_{av} = 144 K Note this includes conduction, convection and radiation heat transfer

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- § Frequently the largest source of heat leak to cryogenic systems
- Fundamental Equation: Stefan-Boltzmann Law energy emitted from an ideal black body: $E_b = \sigma T^4$ where $\sigma = 5.67 \times 10^{-8}$ W/m²K⁴
	- Real world Assumptions:
		- » Emissivity (ε) << 1 and independent of wavelength (grey body)
		- » Two parallel infinite plates: Radiative heat flux (W/m2)

$$
\textsf{Eq. A} \qquad \qquad q_r = \left(\frac{\varepsilon_1 \varepsilon_2}{\varepsilon_1 + \varepsilon_2 - \varepsilon_1 \varepsilon_2}\right) \sigma \left(T_1^4 - T_2^4\right)
$$

» Frequently in cryogenic systems $\varepsilon_1 \sim \varepsilon_2 \ll 1$ then Eq. A becomes:

Eq. B
$$
q_r = \left(\frac{\varepsilon}{2}\right) \sigma \left(T_1^4 - T_2^4\right)
$$

Radiation Heat Transfer

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- » Two long concentric cylinders or concentric spheres (1 represents the inner cylinder): the radiative heat flux (W/m^2) on the inner cylinder is

Eq. C
$$
q_1 = \left(\frac{\sigma (T_1^4 - T_2^4)}{\frac{1}{\epsilon_1} + \left(\frac{A_1}{A_2}\right) \left(\frac{1}{\epsilon_2} - 1\right)}\right)
$$

» Note as is frequently the case in cryogenics, if the spacing between the cylinders is small compared to the inner radius (i.e. $A_1 \sim A_2$) Eq. C becomes Eq. A

Radiation Heat Transfer

- We could reduce the emissivity (ϵ)
	- This is done in some cases; using either reflective tape or silver plating
	- Better below 77 K
	- It's also part of MLI systems (see below)
	- We have to consider tarnishing
	- May be labor intensive

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Radiation Heat Transfer

- § Another way to reduce radiation heat transfer is to install intermediate actively cooled radiation shields that operate at a temperature between 300 K and the lowest cryogenic temperature. This has several advantages.
	- It greatly reduces the heat load to the lowest temperature level » Assume parallel plates with $ε = 0.2$ » then from Eq. B q (300 K – 4.2 K) = 46 W/m² while q (77 – 4.2) = 0.2 W/m²
	- It allows heat interception at higher temperatures & thus better Carnot efficiency
	- Such an actively cooled shield provides a convenient heat intercept for supports, wires etc to reduce conduction heat leak.
- Shields may be cooled by
	- Liquid baths ($LN₂$)
	- Vapor boil off from stored liquid common in LHe storage dewars
	- Cooling flows from refrigeration plants
	- Conductive cooling via small cryocoolers

Examples of Cooled Radiation Shields

- \blacksquare LN₂ bath surrounds inner LHe or $LH₂$ bath
- Baths are separated by a vacuum insulation space
- Shield is cooled by boil off gas from stored cryogen
	- Spacing of cooling tubes on shield may be calculated by: $\Delta T = qL^2/2kt$
		- \sqrt{x} = max allowable temperature difference between any point on shield and tube
		- \rightarrow q = heat flux on shield
		- \triangleright k = shield thermal conductivity
		- \triangleright L = $\frac{1}{2}$ max tube spacing
		- \ast t = shield thickness

Thermal Radiation Shields

- § Uncooled thermal radiation shields placed in a vacuum space between the warm & cold surfaces also help reduce the thermal radiation heat leak
- It can be shown (with the grey approximation and equal emissivities) that with N shields thermal radiation heat transfer is given by:

$$
q = \frac{\varepsilon}{(N+1)2} \sigma \left(T_H^4 - T_L^4 \right)
$$

This is the motivation behind Multilayer Insulation