

# **Lecture 9**

## **Cryostat Design (Part II)**

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# Goals



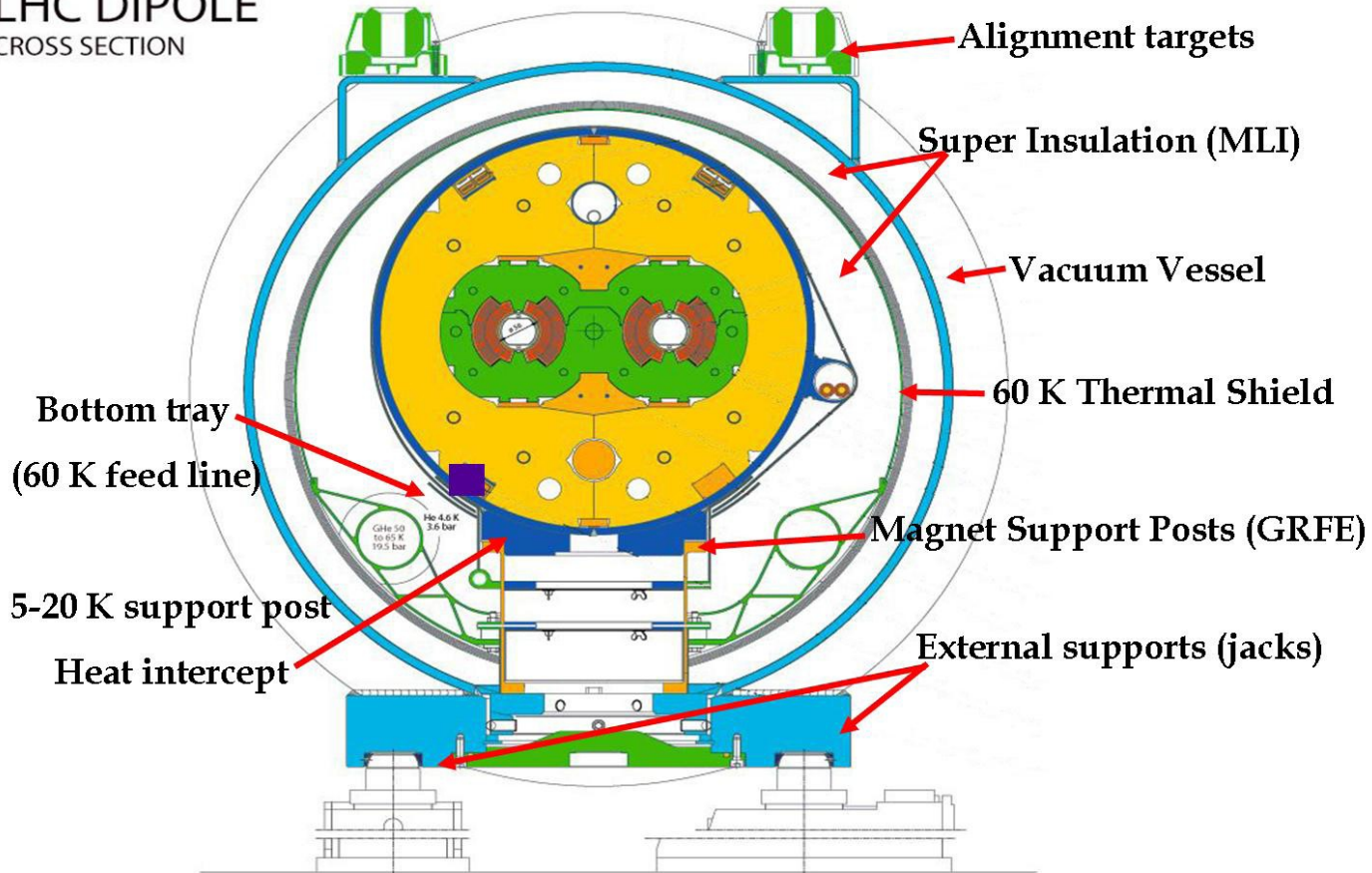
- Describe aspects of cryostat design via example

# Example #1: LHC Dipole Cryostat

- **Some key requirements**
  - 1232 cryostats required
  - High reliability
  - 1.9 K (He II) cooled magnets
  - Low per unit heat leak
  - Minimize cost of materials & assembly
- **Design Choices**
  - Vacuum vessel dimensions chosen to meet industry standard tube sizes
  - GFRE support posts with heat intercepts at 60 K and 5 K
  - Simple integrated aluminum heat shields at 60 K
  - No 5 K radiation shield but MLI blankets between 300 K and 60K and between 60 K and 1.9 K
  - Almost all seals are done via welding with strict QA program
  - Minimal instrumentation
  - Final assembly at CERN via contractor

# LHC Dipole Cross Section

LHC DIPOLE  
CROSS SECTION



“SERIES-PRODUCED HELIUM II CRYOSTATS FOR THE LHC MAGNETS: TECHNICAL CHOICES, INDUSTRIALISATION, COSTS” A. Poncet and V. Parma Adv. Cryo. Engr. Vol 53



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# LHC CROYSTAT COMPONENTS

“SERIES-PRODUCED HELIUM II CRYOSTATS FOR THE LHC MAGNETS: TECHNICAL CHOICES, INDUSTRIALISATION, COSTS” A. Poncet and V. Parma Adv. Cryo. Engr. Vol 53



# LHC Dipole Cryostats

- Cost per unit (not including cold mass)  
~ 100 kCHF (2007 value)  
Compare to 1996 estimate of 130 kCHF (2007 value)
- Largest single cost component is the vacuum vessel at 35.2%
- QA program was 2.4% of unit cost
- Current status: all cryostats delivered, accepted, and cold. LHC currently conducting physics

# Example #2 International Linear Collider SCRF Cryomodule

- Two 15 km long linacs (250 GeV on 250 GeV)
- 35 MV/m SCRF cavities (1.3 GHz)
- Requires ~ 2000 cryomodules
- Extension of TESLA technology
- Requirements are very similar in many ways to LHC dipoles
  - ILC cryomodules have much higher dynamic heat loads

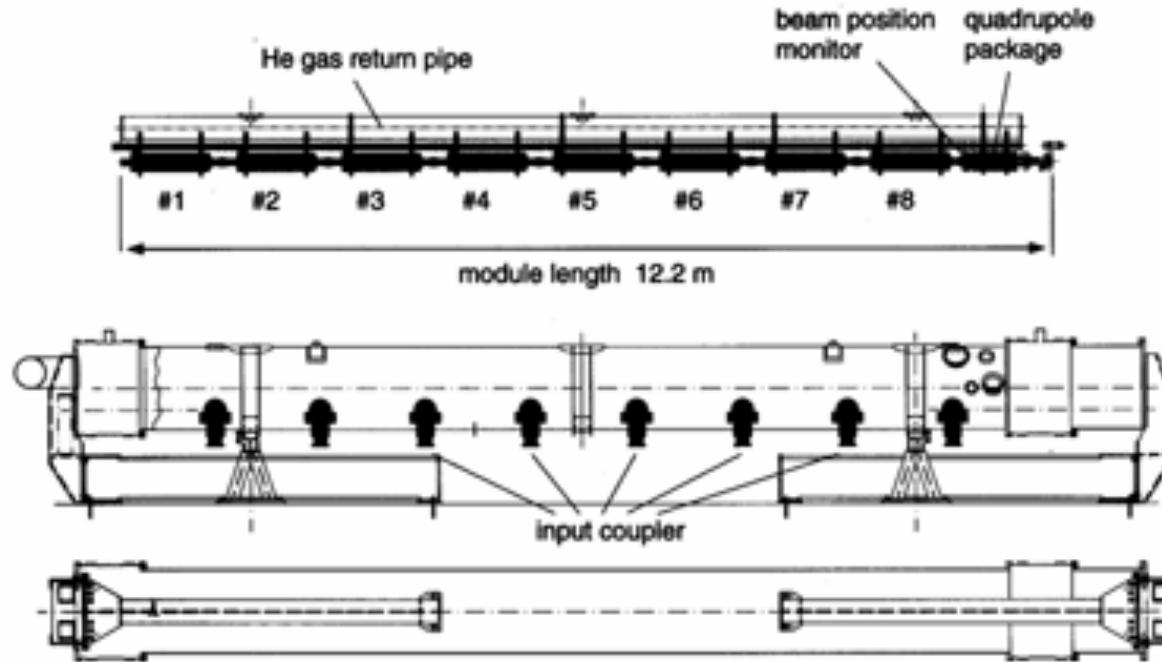


# ILC Cryomodule Features

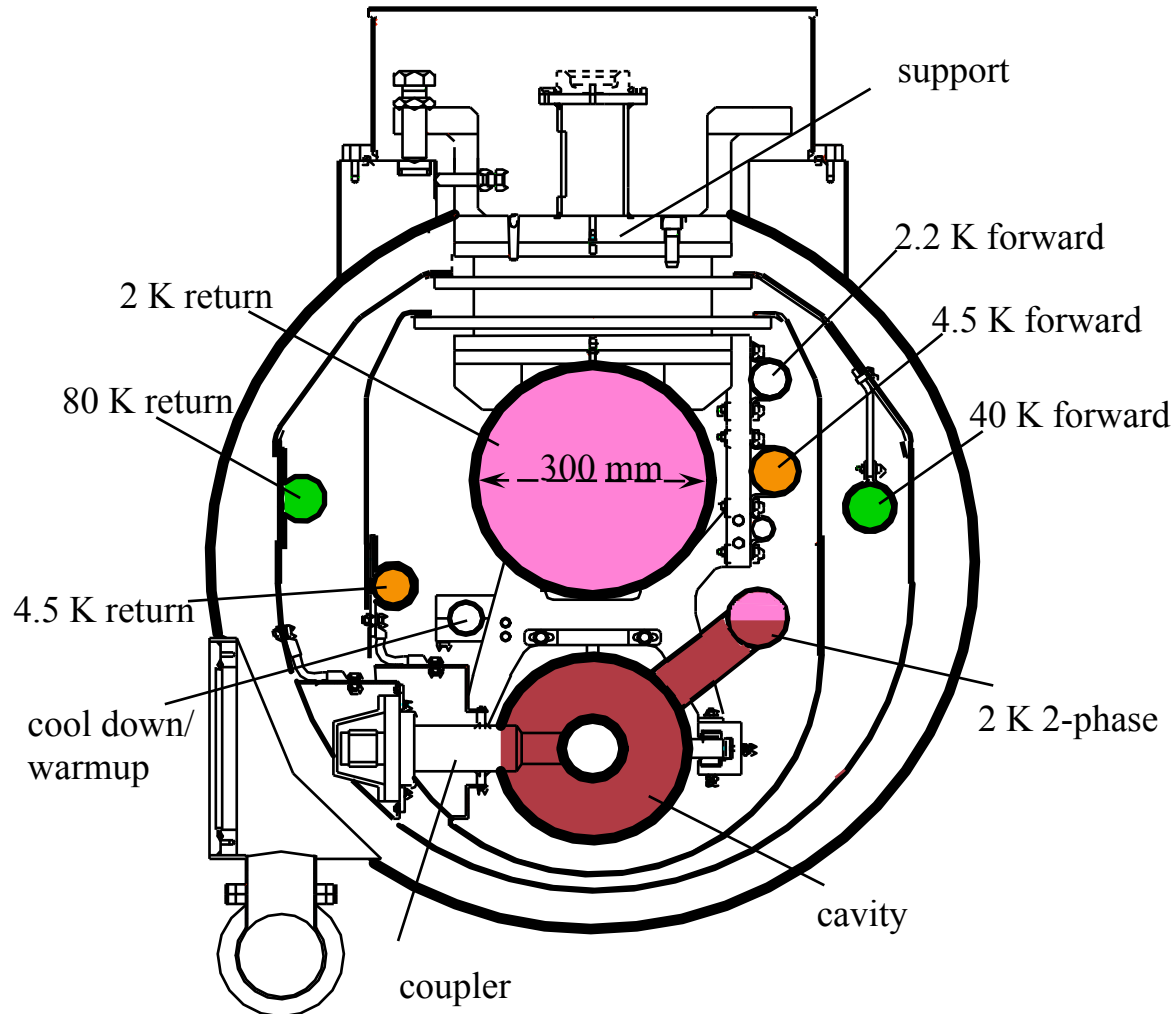
- Eight 9 cell sc cavities + possibly 1 sc magnet package
- Components are tied to 300 mm pipe strongback
- 2 thermal shields (40/80 K and 5 K)
  - 5 K may go away during value engineering
- New design allows semi-fixed couplers
- Design has been extensively tested during the TESLA project
- ILC design is a fourth generation of the TESLA cryomodu



# Side View of 1<sup>st</sup> Generation TESLA Cryomodule (each end of 300 mm tube shrinks 15 mm upon cooldown)



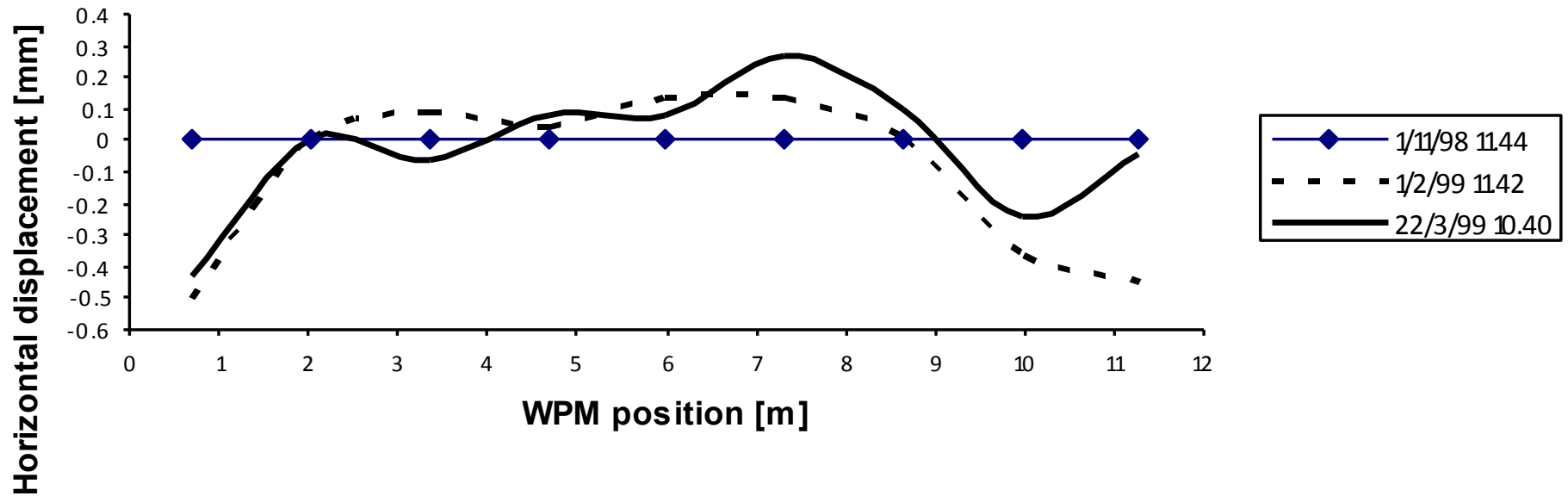
# 3<sup>rd</sup> Generation TESLA Cryomodule ILC Prototype



# TESLA Static Heat Leak Measurements (note total 2 K heat load is ~ 7 W)

Temperature Level	Predicted Heat Leak (W)	Measured Heat Leak (W) Cryomodule #1 (alone)	Measured Heat Leak (W) Cryomodule #1 (with #2)	Measured Heat Leak (W) Cryomodule #2
70 K	76.8	90	81.5	77.9
4.5 K	13.9	23	15.9	13
2 K	2.8	6	5	4

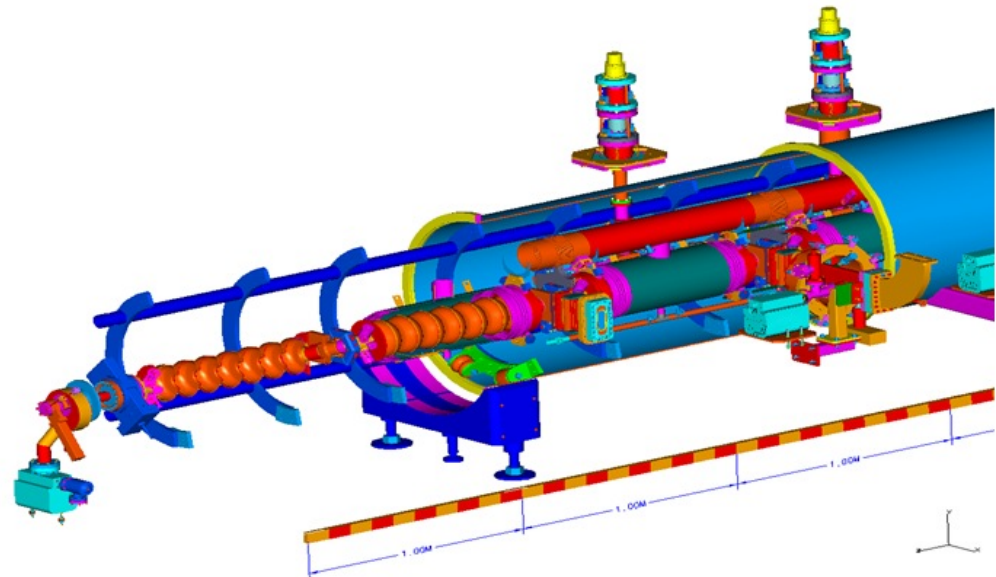
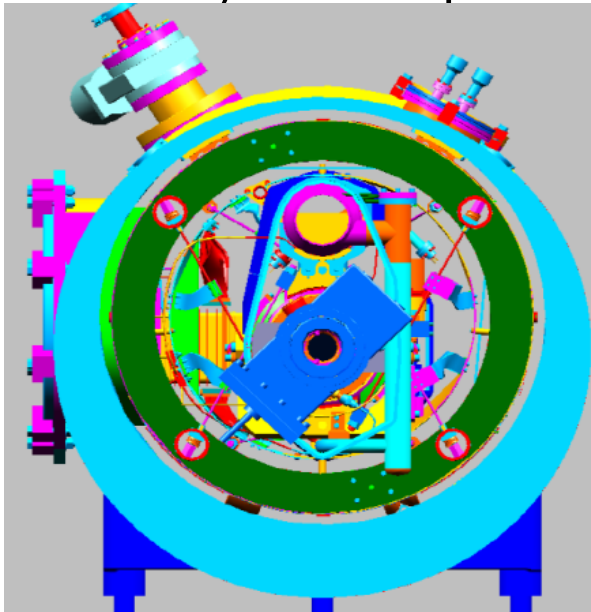
# Alignment Results Cryomodule 2



# Example 3

## JLab 12 GeV Upgrade Cryomodule

- Based on successful SNS design - a total of 10 CMs were needed
- Cavities at 2.1 K, thermal shield at 50 K – no s/c magnets present
- Cold Mass tied to space frame via nitronic rods
- Space frame rolls into vacuum vessel
- Despite extensive SNS experience, design changes (mainly in cold mass) were required after 1<sup>st</sup> prototype – *value of prototyping*

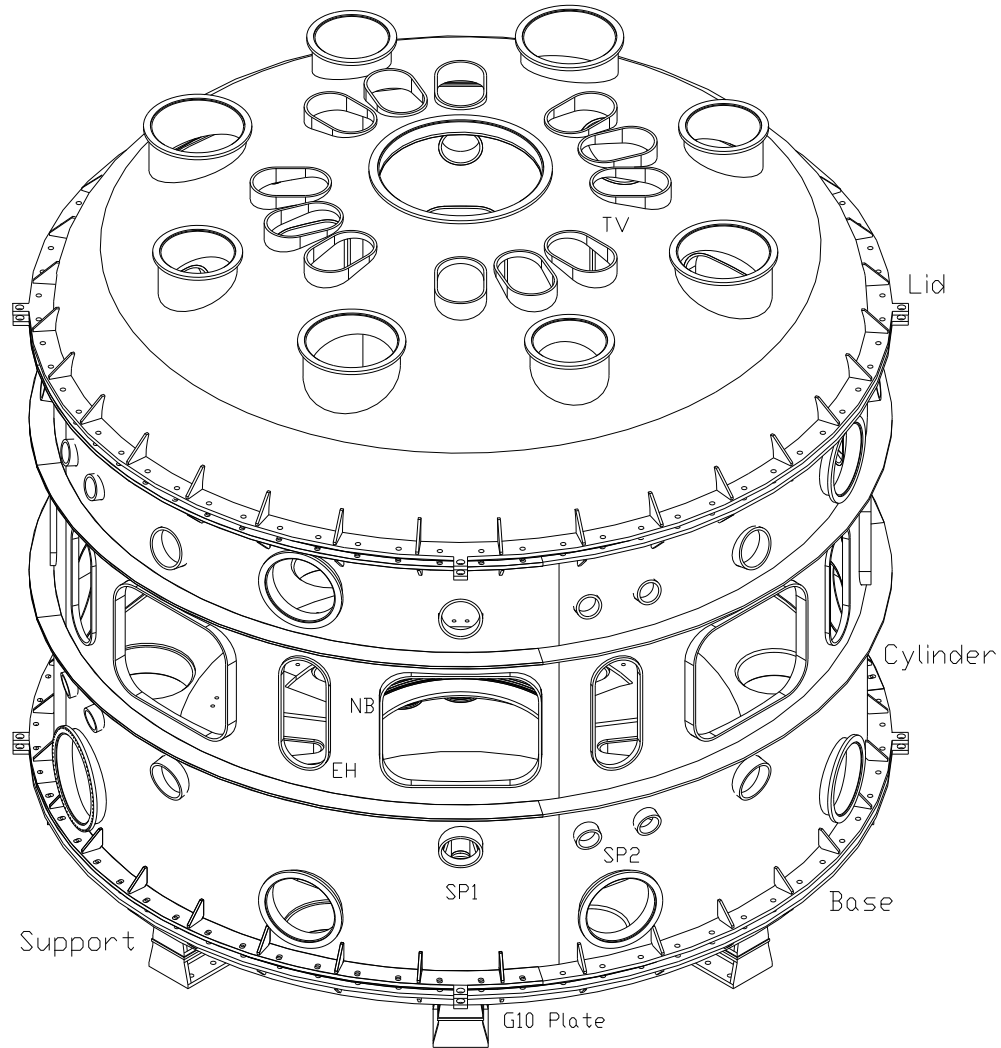




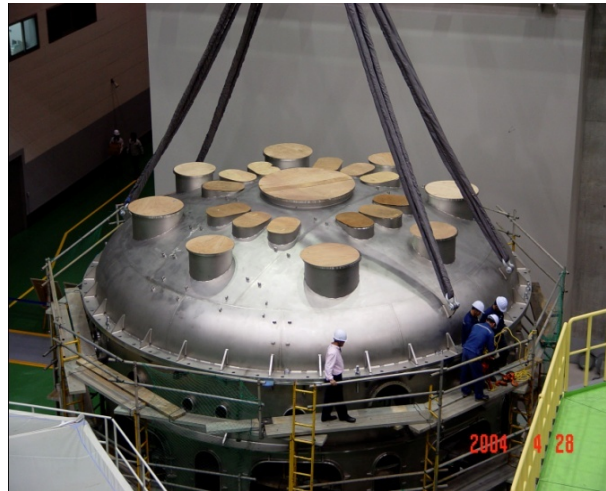
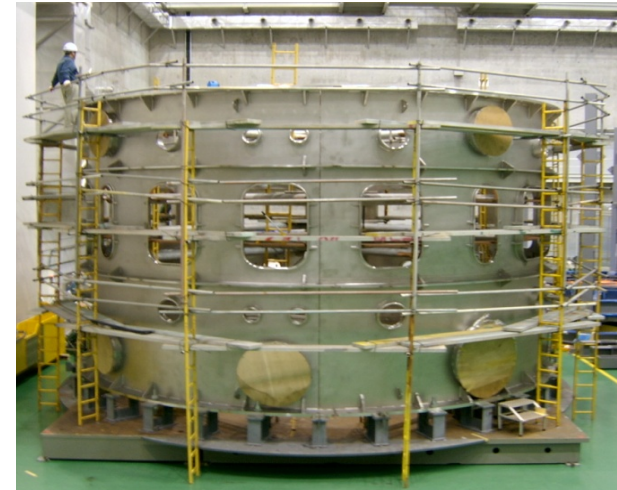
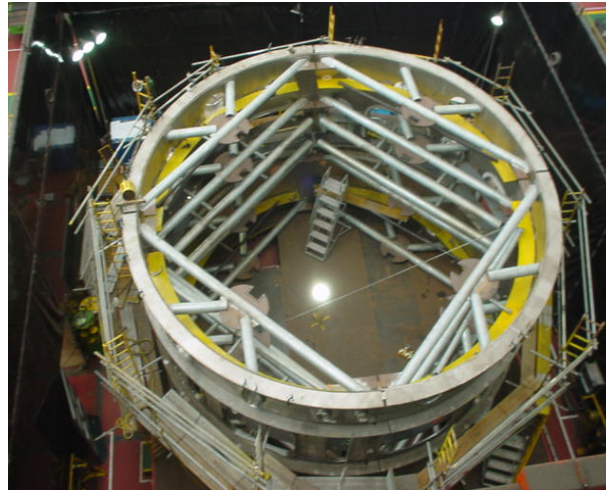
## Example #4 Cryostat and Thermal Shield for KSTAR (Korea Superconducting Tokamak Advanced Research)

- Large (~ 9 m diameter Tokamak using superconducting magnets)
- S/C magnets are contained in a single cryostat (8.8 m ID and 5.7 m high)
- Cryostat is very complex with 72 penetrations
- Designed to ASME pressure vessel code

# KSTAR Vacuum Vessel



# Vacuum Vessel Fabrication






# KSTAR Thermal Shield

- Intercepts thermal radiation from 300 K surfaces
- Actively cooled by He gas to ~ 80 K
- Three classes of shields Cryostat (CTS), Vacuum Vessel (VVTS) and Port (PTS)
  - Space is very limited for VVTS and PTS so MLI blankets are not possible but surfaces were silver plated to improve emissivity

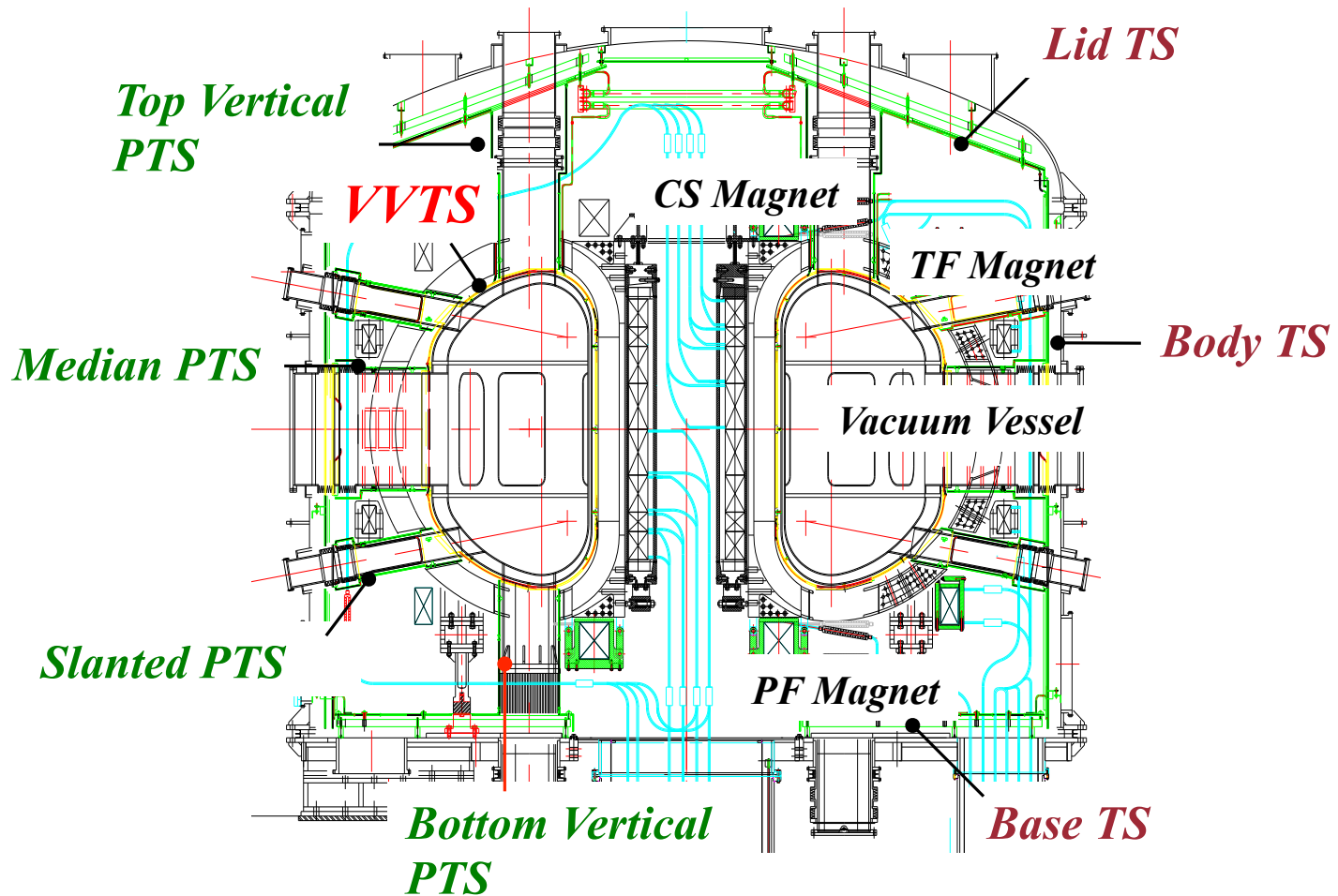
# KSTAR Thermal Shield; Recent Progress

 2007 IAEA

*G. H. Kim, W. C. Kim, H. L. Yang, C. H. Choi, J. S. Bak (NFRC)  
D. K. Kang (Wonshin Eng.)*

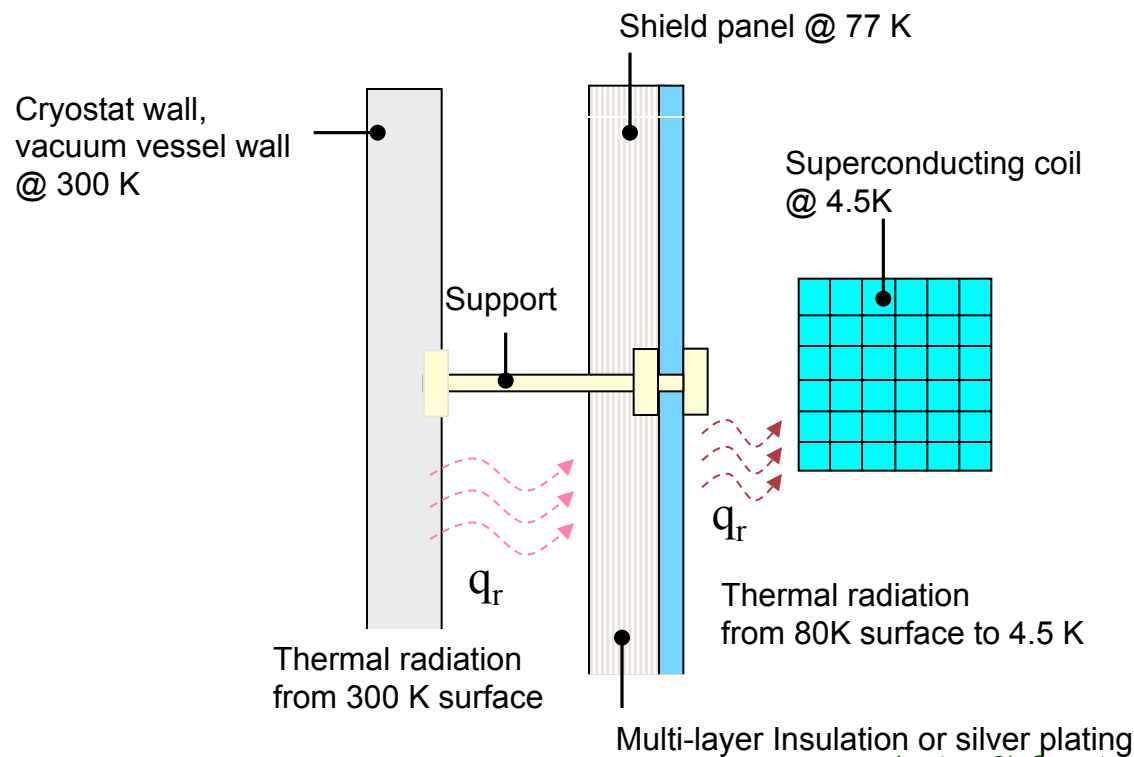
**NFRC** 핵융합연구센터  
National Fusion Research Center

# Thermal Shield Locations



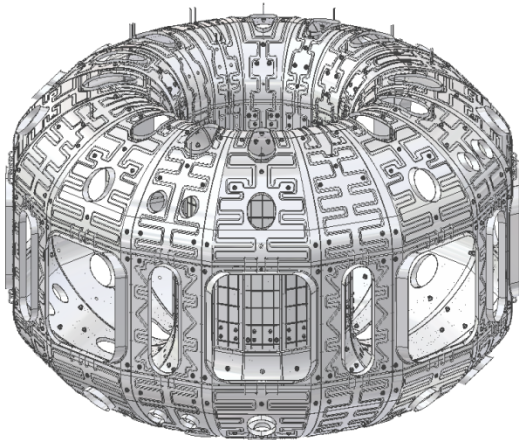
# KSTAR Thermal Shield

- ❑ **KSTAR thermal shield** intercepts radiation from 300 K to 4.5 K and contributes to economic operation of refrigerator system.
- ❑ Consists of three categories: vacuum vessel thermal shield (VVTS), port thermal shield (PTS), and cryostat thermal shield (CTS).
- ❑ Tube on panel configuration is a basic shape of the KSTAR TS.

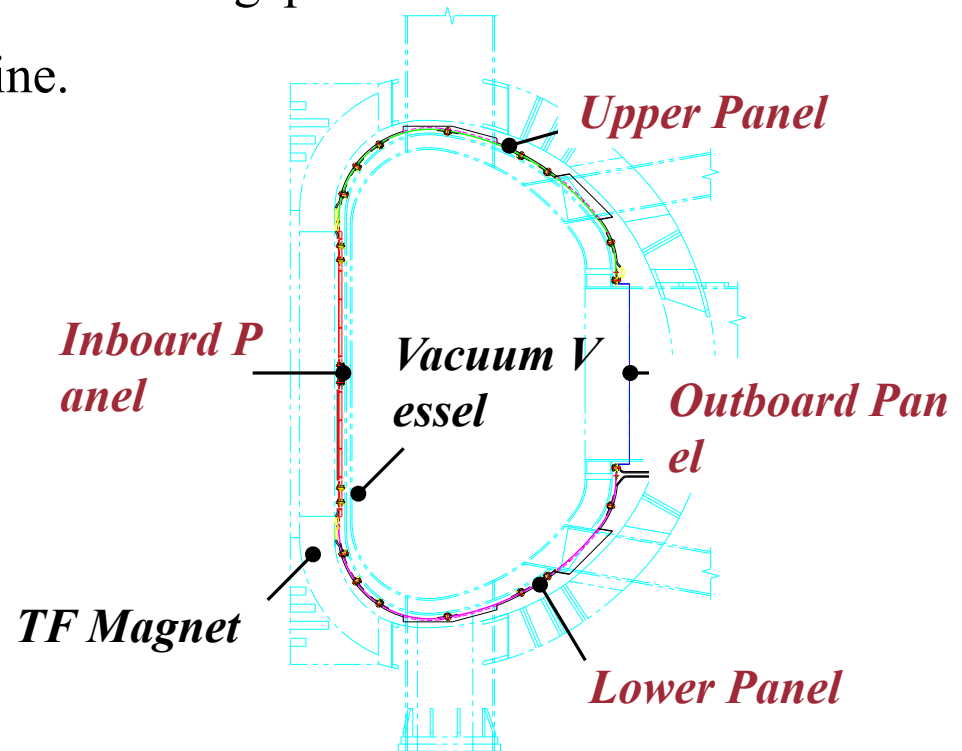


# Vacuum Vessel Thermal Shield

- VVTS is toroidally segmented into 16 sectors and poloidally partitioned into 4 pieces.
- Shield panel is plated by silver of  $10\ \mu\text{m}$  thickness instead of using multi-layer insulation (MLI) due to narrow gap.
- VVTS has redundancy cooling line.



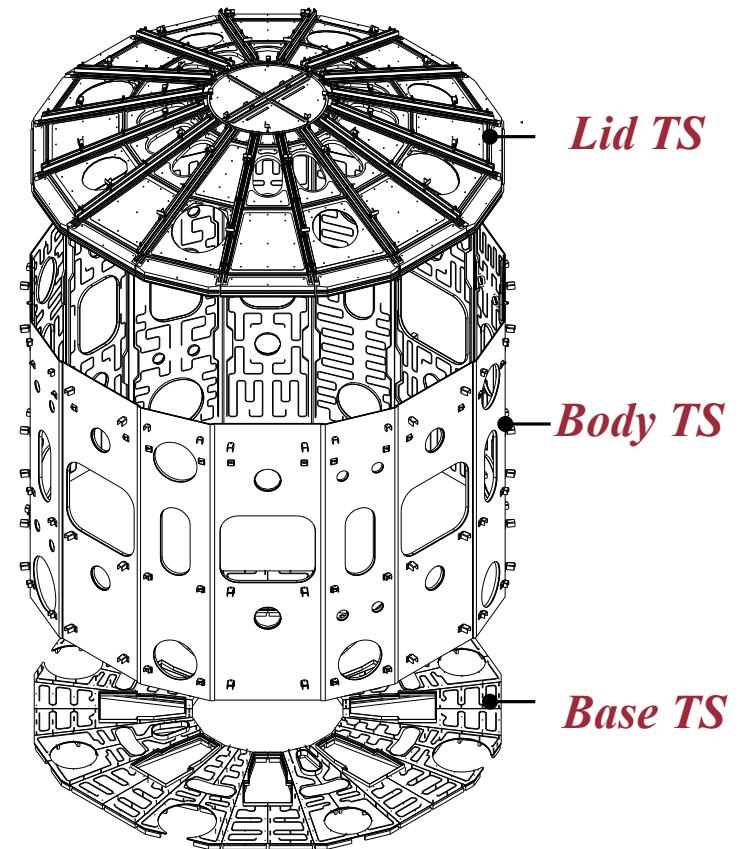
Isometric view of vacuum vessel thermal shield



Four sub-sectors in poloidal direction

# Cryostat Thermal Shield

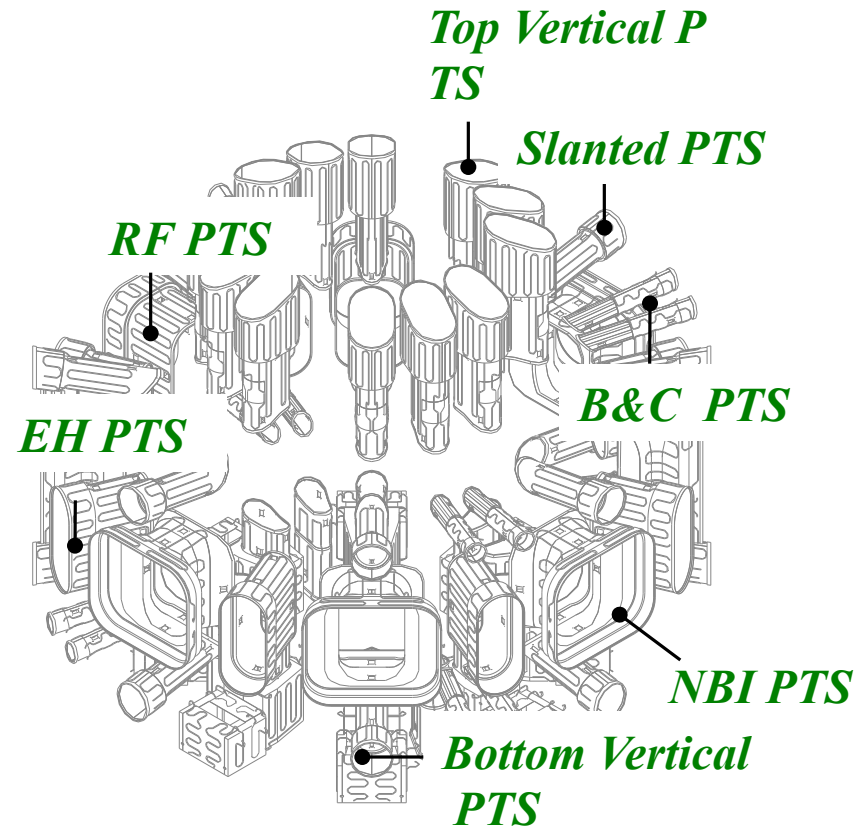
- ❑ CTS is placed on the cryostat inner surface wall.
- ❑ MLI is adopted to prohibit thermal radiation from cryostat surface.
- ❑ Each part is toroidally divided into 16 sectors.
- ❑ Shield panel consists of a flat stainless plate, MLI of 30 layers, 1 mm thick stainless steel sheet, and G10 spacers.



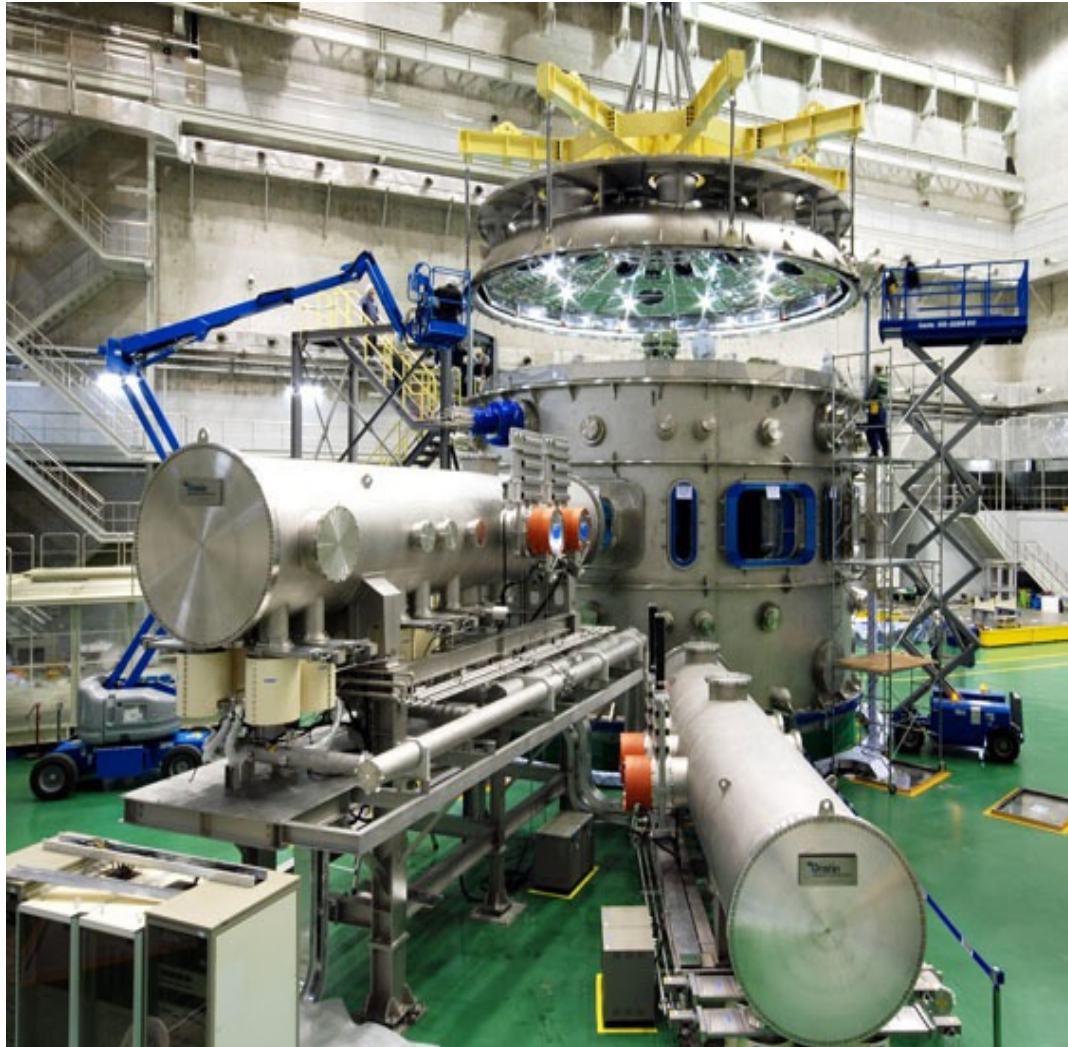
Exploded view of cryostat thermal shield

# Port Thermal Shield

- There are 72 penetration ports for connecting the vacuum vessel body and cryostat.
- Ports are classified into 7 types:  
NBI, EH, RF, SL, B&C, TV, BV
- port.
- **PTS** covers all of ports.
- Silver plating method is adopted for the PTS.
- All PTSs are divided into two parts: lower and upper part.



Isometric view of port thermal shields



**KSTAR Thermal Shield; Recent Progress**  
**G.H. Kim, W.C. Kim, H.L. Yag, C.H. Choi, J.S.Bak, D.K. Kang**

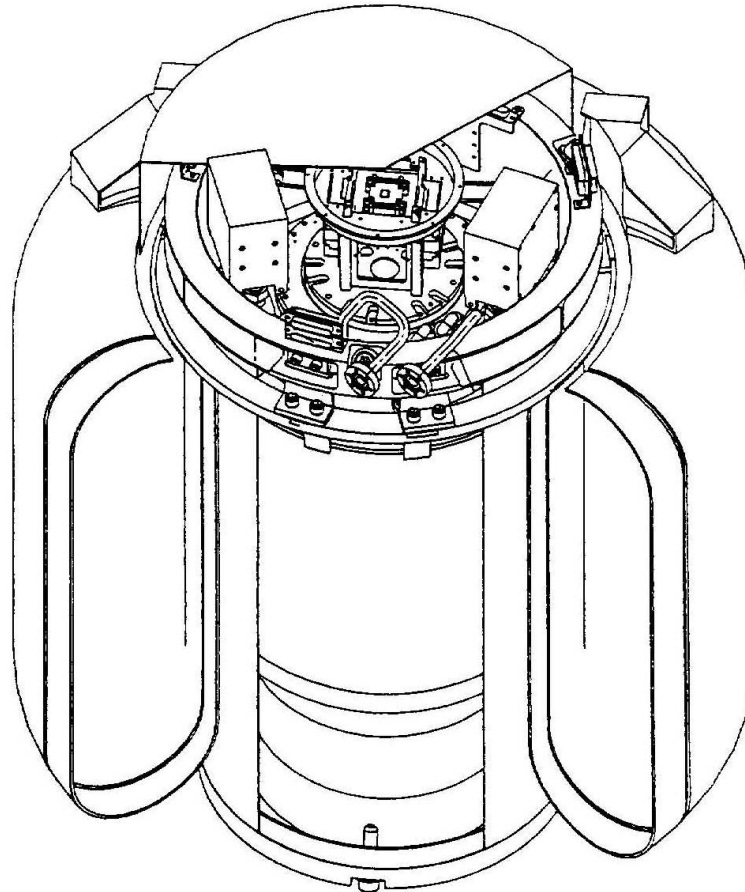


# Example #5 Space Cryostat The X-Ray Spectrometer (XRS)

- Mission life time is dependent on the supply of He II (1.3 K) in the cryostat. In order to achieve the 2.5 year life time, the heat leak must be  $< 800 \mu\text{W}$
- Additional design drivers:
  - Size and weight
  - Use of an ADR requiring a superconducting magnet – sensors need 0.065 K
  - A costly one of a kind device

- All heat leaks (even  $10 \mu\text{W}$ ) are important
- Solid Ne (17 K) dewar surrounds He dewar to reduce radiation and conduction heat leak
- Low emissivity materials used: polished Al, gold plating and aluminized mylar
- $\text{HiT}_c$  superconductors used for magnet leads
- All other wiring is optimized for minimum heat leak
- Helium tank suspended by graphite/epoxy straps optimized to meet launch loads
- Radiation baffles in vent and fill lines plus devices to prevent superfluid film flow in vent line

“Thermal Design of the XRS Helium Cryostat”,  
S. Breon et al., Cryogenics 36:10 (1996)



**Figure 1** XRS configured for Astro-E. Hybrid cryogenic system provides cooling stages at 17 K (solid neon), 1.3 K (superfluid helium) and 0.065 K (ADR). JFETs in FEA are at 80–120 K

# “Thermal Design of the XRS Helium Cryostat” S. Breon et al., *Cryogenics* 36:10 (1996)

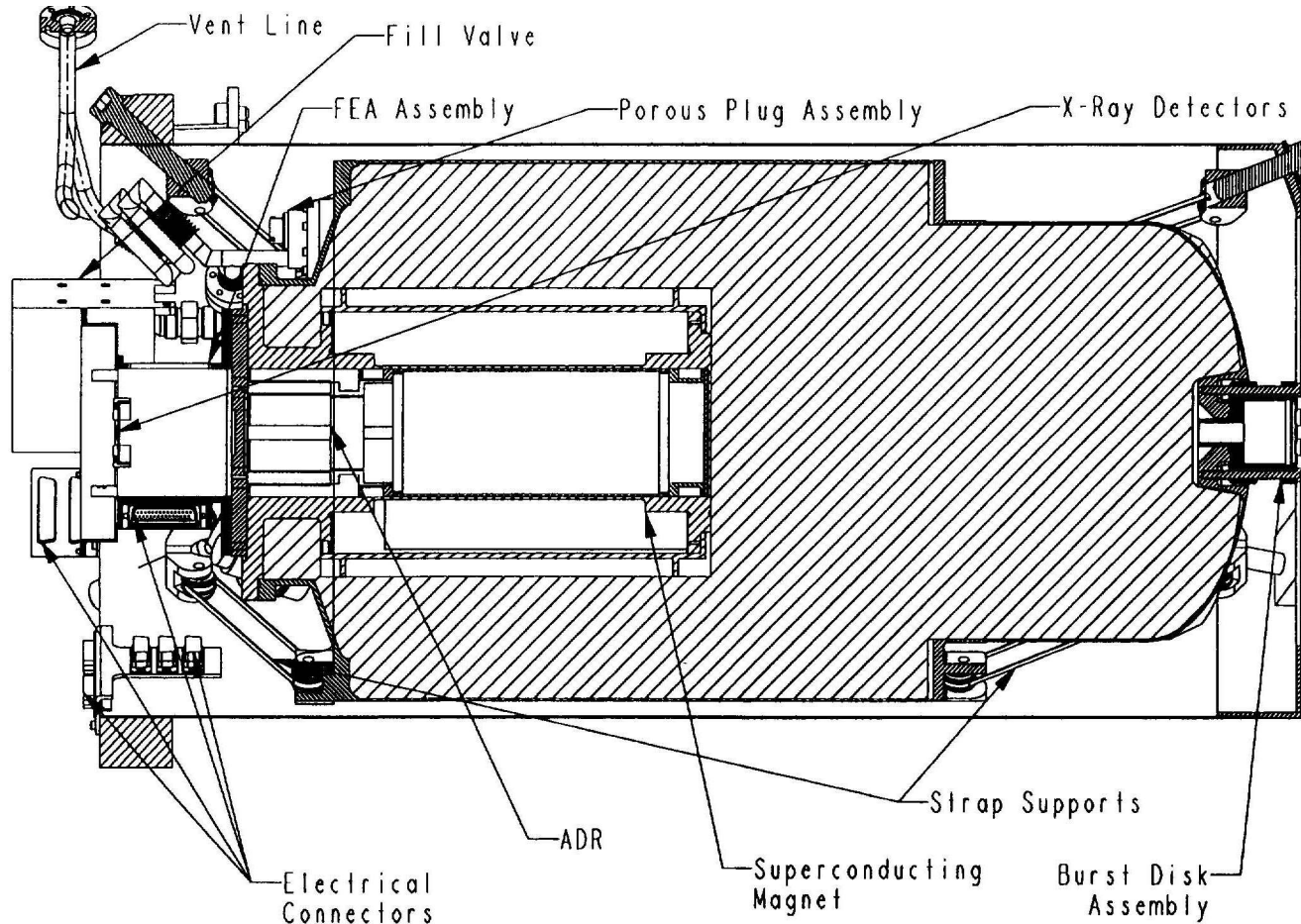


Figure 2 XRS helium cryostat



# XRS Cryostat

- Analytical heat leak models and full scale thermal measurements at the component, subsystem and helium insert level were carried out.
- Measured heat leak to the helium insert (on the ground) was  $\sim 629 \mu\text{W}$

- Define requirements first
- Design in safety from the start
- Use appropriate materials for cryogenic temperatures
- Review literature & learn from previous efforts
- Use tested commercial solutions whenever possible
- Avoid feedthroughs & demountable seals at cryogenic temperatures
- Conduct prototype tests when required